Structural Analysis of Visco Fan Insert for Automobile Engine Cooling System using Finite Element Method

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Abstract- A heavy duty automobile has a cooling module with big fan driven by engine (mounted on crank shaft or driven through belt-pulley) for cooling air flow over the cooling module. As heavy duty vehicle required more traction power hence the engine also big. So reduce consumption of energy by the fan, it has to operate with visco clutch driven by engine itself, the running and stopping of fan occurs many times instead of continuous running during the engine in operation. The load on the fan is thrust due to air flow over it. There is external excitation from the engine or from the bearing unbalance during the assembly. To reduce the weight and cost such a fan can be made of plastic with hub made with higher stiffness to sustain jerk during the starting and stopping of fan during operation. Fan insert is usually made of steel because it is easily put while plastic fan-blade is manufactured by injection molding process. Fan inserts can be directly inserted while injection molding of fan blade. Fan insert is important part of Cooling Visco Fan system as the fan supports the engine cooling or keeping specific engine operating temperature and thus keeping the vehicle on the road. Thus, it is necessary for the designers to provide a better design of parts having maximum reliability under all loading conditions. FEA approach can be applied for the calculation of strength and to check possibility of part failure. 3D model of a fan will be drawn in CATIA V5R19, Meshing will be carried out in MEDINA, and PERMAS will be used for solutions. Model will be fabricated and testing will be done on vehicle or test lab.

Keyword—Visco Fan insert, design Optimization, Design of insert.

I.INTRODUCTION

A Visco fan insert is part of an automobiles engine cooling system. It is located in front of the vehicle and it is in between engine cooling system and engine. It is mounted on directly on Visco clutch. The Visco clutch is driven by engine shaft directly or by means of belt-pulley arrangement.

In older days there was use of rigid fan which is directly mounted on the crank shaft which causes over cooling of the engine or decrease in the engine efficiency to overcome this? Problem we use visco-clutch to drive fan "VISCO-CLUTCH" VISCO stands for viscous behavior of operating oil and CLUTCH for engagement and disengagement of clutch parts. Position Of Visco-clutch Fan

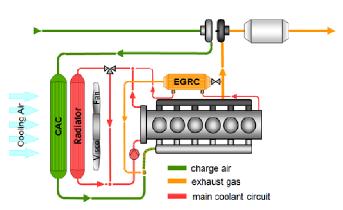


Fig.1.1 Position of Visco-clutch Fan

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A Fan-insert is an automobile engine cooling module design in which one or more heat exchangers are mounted in front of the engine (and perpendicular to and forward of) the driven direction of the vehicle.

Clutch is defined as "The part which transmits rotational motion of one shaft to other shaft". The rotating s visco clutch (spindle) and gyroscopic effects creates a forced vibration problem [1].

The consideration for no damping measures are a logical for undamped systems which generally work well for systems where the damping is too less [5].

II.METHODOLOGY

It is extremely important for a engine cooling module system to have better functionality and life. The engine cooling system must work reliably under all working conditions. Poor strength properties of the engine cooling system limit the life expectancy of the vehicle. Therefore, engine cooling system parts must be strong enough to cope with all working conditions.

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But the existing design has already set with respective to the required air flow hence we could not change the parameter of fan blade. In our project, we are going to work over fan insert of a visco-fan. By reverse engineering (hand calculation) dimensions of the fan will be found out of existing design. 3D modeling will be done in CatiaV5. Meshing and analysis will be carried out in MEDINA and PERMAS respectively. Stress and deformation will be the output of analysis. By knowing the low & high stressed region, material from that region can be removed & added; re-designing the fan insert will be done, following the same procedure for 3D modeling, meshing and analysis. Stress values must be below critical value to ensure that the new design is safe.

Our main motto is to optimize the design of fan inserts So that it will not fail during its operations. On most models, Fan inserts is fails/cracks at the mounting points leading to dangerous consequences like failure of engine due to inefficient/lower performance of engine cooling system. There are numerous reports of the visco fan insert failure. If this should happen at highway speeds you could lose control of the vehicle due to failure of engine due to inefficient cooling. In this paper we will do the reverse engineering and get the exact dimensions of our part. We will make 2D model of our part using CATIA V5R19. Afterwards we will do meshing of that model. The static and dynamic loads within the boundary conditions are applied. The meshing of the model using MEDINA 8.5.1.3 is done. The PERMAS V15 is used as solver to get solutions of this meshed model. Re-Sequence of the above procedure with new optimized design and Result comparison.

CAD MODEL

Prepare a CAD model from input parameters: sketch, photographs and hand measurements, using 3D modeling software like CATIA Tool. Based on input parameter form Company for fan insert, shaft and bearing CAD modeling has carried out.

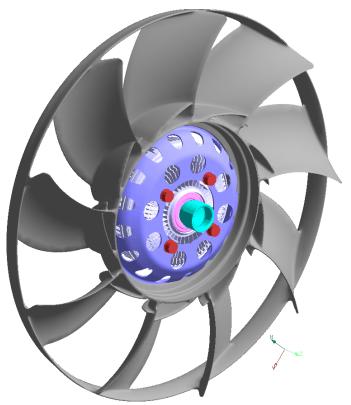


Fig.2.1 CAD model of Visco Fan

Computer-aided design (CAD), and it is also known as computer aided design and drafting (CADD). The process of design and design documentation is done with the help of this computer technology. The process of drafting is done by Computer Aided Drafting with a computer. For the purpose of streamlining design processes; drafting, documentation, and manufacturing processes, users are provided by CADD software or environments and with input-tools. The detailed engineering of 3D models and/or 2D drawings of physical components are done mainly with the help of CAD, but it is also used in the complete engineering process from conceptual design and layout of products, through strength and dynamic analysis of assemblies to definition of manufacturing methods of components. It can also be used to design objects.

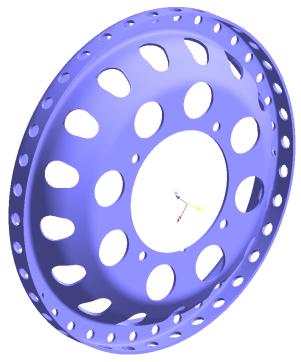
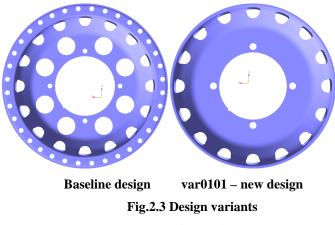


Fig.2.2 Visco Fan Insert

Design Variant:

With the baseline design the component has tested and it fails due to high stresses at fan blade insert, hence design modifications are done. Two new design variants of fan insert are simulated



var0101 - The holes are removed in baseline

var0201 - Thickness is increased by 25% in Var0101

A General Procedure for (FEM) Finite Element Analysis:

The steps in formulating a finite element analysis of a physical problem are common to all analyses; It may be structural, heat transfer, fluid flow, or some other problem. These steps are included in commercial finite element software packages. Some of the steps are described as follows.

1) Preprocessing

The preprocessing step is, quite generally, described as defining the model and includes

- Define the geometric domain of the problem.
- Define the element type(s) to be used.

There are different types of elements for discretization. Fig.3.2 shows the different types of elements used for meshing.

- 1) 1D Element: Rod, Bar, Beam
- 2) 2D Element: Shell, Membrane, Plane stress, Plane strain
- 3) 3D Element: Solid

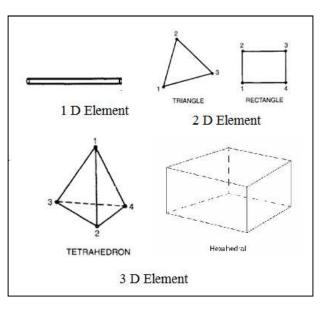


Fig.2.4 Different types of elements

- Define the material properties of the elements.
- Define the geometric properties of the elements (length, area, and the like).

After meshing, material (e.g. Young's Modulus) and property information (e.g. thickness values) are assigned to the elements. In Model, basically Fan blade is made up of plastic material, Visco clutch body is made of aluminium and fan insert is made up of Steel material. The material and their properties are shown in table 4.1.

Material	Young's	Passions ratio	Density	
	modulus(E)	(µ)	(tonne/ mm ³)	
	(N/mm2)			
Plastic	6,100	0.35	1360× 10 -9	
Steel	210,000	0.38	7.8× 10 ⁻⁹	
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• Define the element connectivity's (mesh the model).

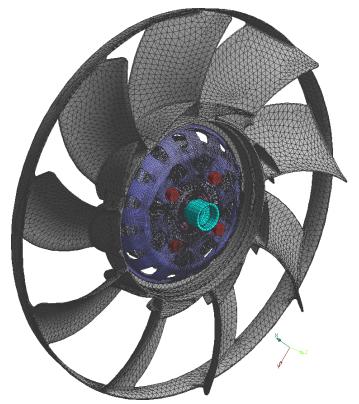


Fig.2.5 Mesh model of Visco fan

• Define the physical constraints (boundary conditions). The shaft of the visco fan (shaft) is fixed in all directions except rotation about its own axis is free for centrifugal load and fixed for thrust load.

The thrust force is more (10-20%) at tip of the fan blade [6].

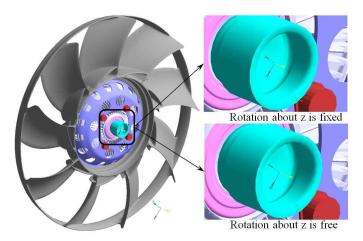
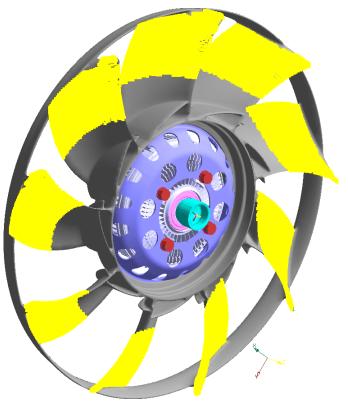


Fig.2.6 Constrains for static simulations

Define the loadings.

•

In linear static analysis, centrifugal load due to its own weight and the thrust force due to air pushed by fan at N_{max} rpm are considered as load condition. After applying the boundary condition in the preprocessor the model is solved in the Permas. The results of solved model are evaluated in MEDINA post-processing.



Thrust is applied at yellow mark region of fan blade Fig.2.7 Load for static simulations

• The preprocessing (model definition) step is critical.

2) Solution-:

During this solution phase, finite element software assembles the governing algebraic equations in matrix form and then using known parameters computes the unknown values of the primary field variable(s). The additional, derived variables, such as reaction forces, element stresses are computed with known results values using back substitution. Special solution techniques are used to reduce data storage requirements and computation time to solve the thousands of equations. Based on Gauss elimination, a wave front solver is commonly used for static linear problems.

3) Post processing

Analysis and evaluation of the solution results is referred to as post processing. Postprocessor software contains sophisticated routines used for sorting, printing, and plotting selected results from a finite element solution. Examples of operations that can be accomplished include-:

- Sort element stresses in order of magnitude.
- Check equilibrium.
- Calculate factors of safety.
- Plot deformed structural shape
- Animate dynamic model behavior.
- Produce color-coded temperature plots.
- While solution data can be manipulated many ways in post processing, the most important objective is to apply sound engineering judgment in determining whether the solution results are physically reasonable.

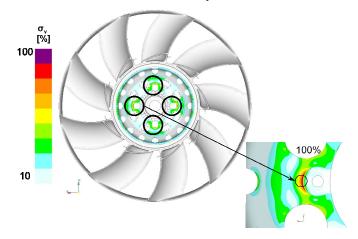
III.RESULTS AND DISCUSIONS

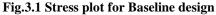
Static Analysis

Engine operating conditions are considered which have maximum rotational speed of N_{max} rpm is applied on the fan in linear static analysis. The maximum stress location due to centrifugal force does not match with failure in test whereas that is match with maximum stress location due to thrust. The maximum stress due to centrifugal force is (70%) less as compared to that of due to thrust force at N_{max} rpm.

Static Analysis Results	Baseline	var0101	var0201
Centrifugale force @ N _{max} rpm	100%	98%	94%
Thrust force @ N _{max} rpm	100%	74%	25%

Table 3.1 Static analysis results





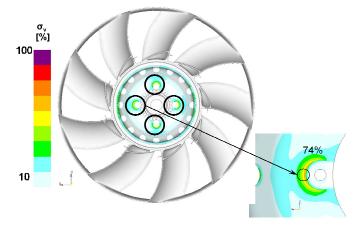


Fig.3.2 Stress plot for design var0101

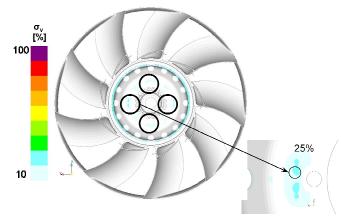


Fig.3.3 Stress plot for design var0201

From the static analysis results we concluded that the stress in the fan inserts are less as compared to baseline (failure) for var0101 & var0201.

Dynamic Analysis

Structure's vibration characteristics such as natural frequencies, mode shapes, and mode participation factors can be determined by Modal analysis technique [4].

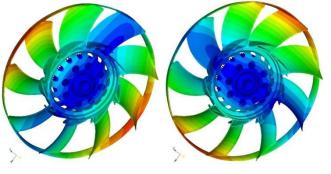
The Eigen (natural frequency) values and modes are calculated with rotations (N_{max} rpm) and without rotations. The Eigen values for both the design as follows.

Modes	Eigen Frequency					
widues	Baseline		Var0101		var0201	
	0 rpm	N _{max} rpm	0 rpm	N _{max} rpm	0 rpm	N _{max} rpm
1	100%	108%	107%	115%	121%	138%
2	100%	108%	107%	115%	130%	137%
3	100%	137%	101%	138%	109%	141%
4	100%	112%	100%	112%	106%	117%
5	100%	111%	101%	112%	106%	117%
6	100%	123%	101%	123%	103%	124%

Table 3.1 Eigen/Natural Frequency at no rotation and

at rpm

The modes for baseline without rotation (0 rpm) are as follows. The mode 3 will be responsible of fan insert bending.



Mode 1

Mode 2

Fig.3.4 Modes 1 & 2 of baseline

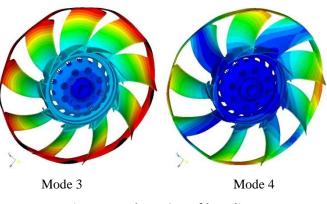


Fig.3.5 Modes 3 & 4 of baseline

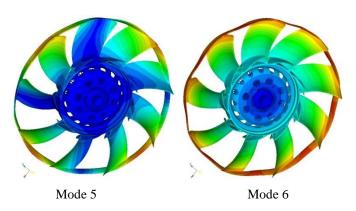


Fig.3.6 Modes 5 & 6 of baseline

The Campbell diagram gives us the information about the change in natural frequencies with rotational speed due to gyroscopic stiffness variation. The Campbell diagram for baseline design is calculated in PERMAS and it is as shown below.

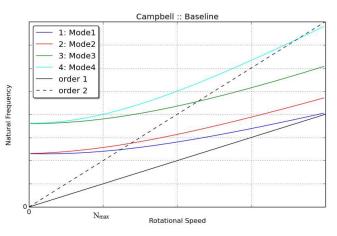
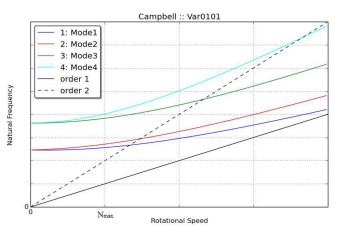


Fig.3.7 Campbell diagram of Baseline Design





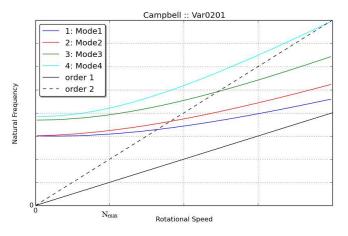


Fig.3.9 Campbell diagram of Design var0201

The design var0101 has small influence on the natural frequencies where as design var0201 has big influence. The first and second modes are shifted by 7% in design var0101 whereas those are shifted 21-30% in design var0201.It means the effect of natural frequency on operating speed range is more for baseline and it decreases with design var0101 and var0201.

IV.CONCLUSION

The New structure (design var0201) is performing better with satisfying loading conditions. The reduction in high stress will lead to performed better as compared to existing design. Overall stress concentration and deformation is less for var0101 & var0201.On safer side design var0201 is manufactured and tested which shows no failure at the end of test.

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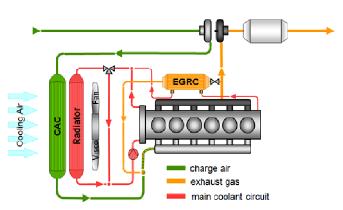


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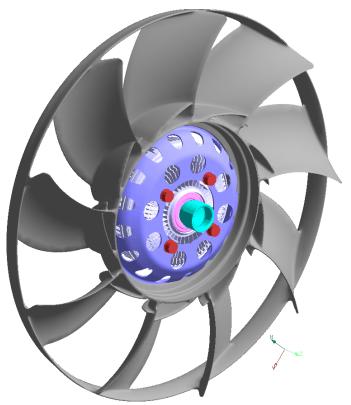


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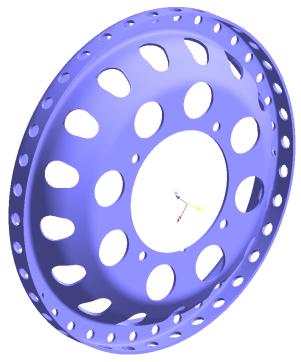
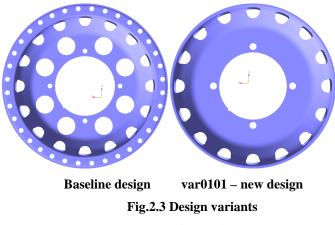


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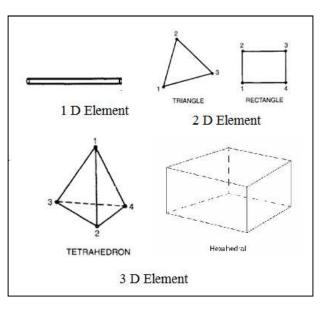


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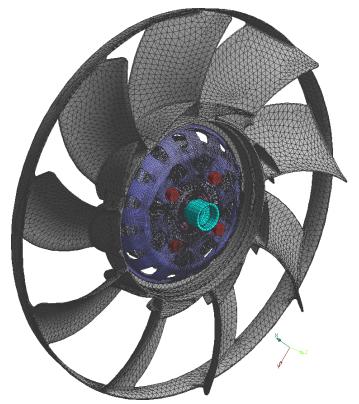


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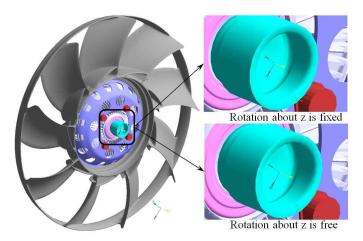
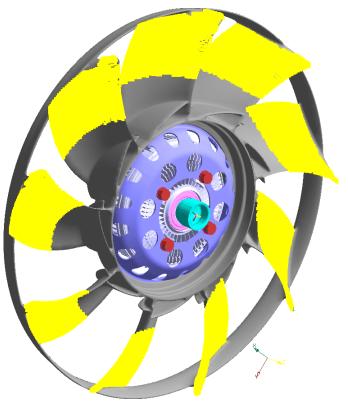


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- Produce color-coded temperature plots.
- While solution data can be manipulated many ways in post processing, the most important objective is to apply sound engineering judgment in determining whether the solution results are physically reasonable.

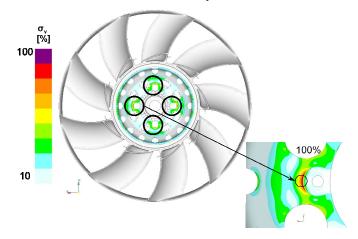
III.RESULTS AND DISCUSIONS

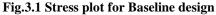
Static Analysis

Engine operating conditions are considered which have maximum rotational speed of N_{max} rpm is applied on the fan in linear static analysis. The maximum stress location due to centrifugal force does not match with failure in test whereas that is match with maximum stress location due to thrust. The maximum stress due to centrifugal force is (70%) less as compared to that of due to thrust force at N_{max} rpm.

Static Analysis Results	Baseline	var0101	var0201
Centrifugale force @ N _{max} rpm	100%	98%	94%
Thrust force @ N _{max} rpm	100%	74%	25%

Table 3.1 Static analysis results





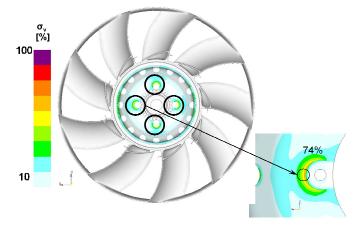


Fig.3.2 Stress plot for design var0101

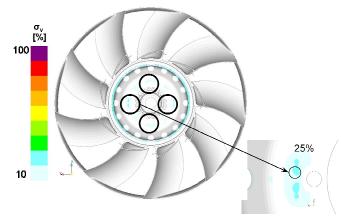


Fig.3.3 Stress plot for design var0201

From the static analysis results we concluded that the stress in the fan inserts are less as compared to baseline (failure) for var0101 & var0201.

Dynamic Analysis

Structure's vibration characteristics such as natural frequencies, mode shapes, and mode participation factors can be determined by Modal analysis technique [4].

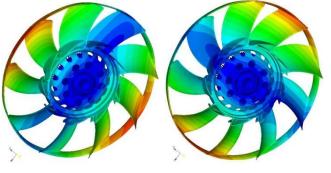
The Eigen (natural frequency) values and modes are calculated with rotations (N_{max} rpm) and without rotations. The Eigen values for both the design as follows.

Modes	Eigen Frequency					
widues	Baseline		Var0101		var0201	
	0 rpm	N _{max} rpm	0 rpm	N _{max} rpm	0 rpm	N _{max} rpm
1	100%	108%	107%	115%	121%	138%
2	100%	108%	107%	115%	130%	137%
3	100%	137%	101%	138%	109%	141%
4	100%	112%	100%	112%	106%	117%
5	100%	111%	101%	112%	106%	117%
6	100%	123%	101%	123%	103%	124%

Table 3.1 Eigen/Natural Frequency at no rotation and

at rpm

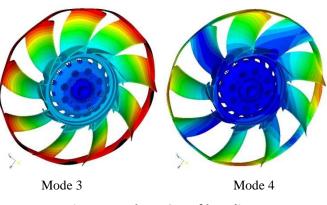
The modes for baseline without rotation (0 rpm) are as follows. The mode 3 will be responsible of fan insert bending.

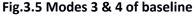




Mode 2

Fig.3.4 Modes 1 & 2 of baseline





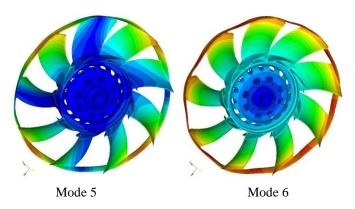


Fig.3.6 Modes 5 & 6 of baseline

The Campbell diagram gives us the information about the change in natural frequencies with rotational speed due to gyroscopic stiffness variation. The Campbell diagram for baseline design is calculated in PERMAS and it is as shown below.

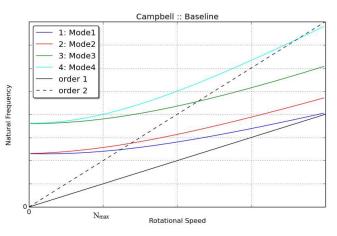
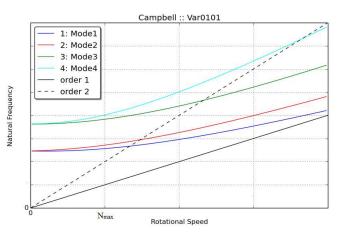


Fig.3.7 Campbell diagram of Baseline Design





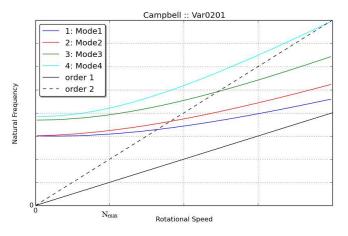


Fig.3.9 Campbell diagram of Design var0201

The design var0101 has small influence on the natural frequencies where as design var0201 has big influence. The first and second modes are shifted by 7% in design var0101 whereas those are shifted 21-30% in design var0201.It means the effect of natural frequency on operating speed range is more for baseline and it decreases with design var0101 and var0201.

IV.CONCLUSION

The New structure (design var0201) is performing better with satisfying loading conditions. The reduction in high stress will lead to performed better as compared to existing design. Overall stress concentration and deformation is less for var0101 & var0201.On safer side design var0201 is manufactured and tested which shows no failure at the end of test.

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